

Spacecraft anatomy: bus, payload and subsystems

Essential answer

Spacecraft anatomy: bus, payload and subsystems. The question to solve is: When we say “the spacecraft”, what machines are we really talking about? From far away an Earth-Mars vehicle looks like one object. In reality structure, power, thermal, avionics, communications, GNC, propulsion and payload must work together, sometimes for months. The rest of the course turns that idea into an auditable line of reasoning: explicit units, stated assumptions, reproducible calculations, order-of-magnitude checks and interpretation limits. A result is useful only when the reader can explain what it measures, where every input came from and which engineering decision it can support.

Reasoning map

- 1. 1 — The concrete scene
- 2. 2 — Essential words, explained before using them
- 3. 3 — See the architecture before calculating
- 4. 4 — Formulas, only when they answer a question
- 5. 5 — What units and margins mean

From far away an Earth-Mars vehicle looks like one object. In reality structure, power, thermal, avionics, communications, GNC, propulsion and payload must work together, sometimes for months.
Shared support functions enabling the mission.

Before calculating

Quantity:	Spacecraft anatomy: bus, payload and subsystems
Check:	units + order of magnitude + assumptions
Decision:	state what the result actually supports

Spacecraft anatomy: bus, payload and subsystems

2 — Essential words, explained before using them

- Shared support functions enabling the mission.
- What the vehicle carries to accomplish its purpose.
- Organised family of functions and hardware.
- Boundary exchanging data, power, heat, loads or matter.
- Verifiable condition imposed on design.
- Reserve between need and capability.

3 — See the architecture before calculating

- Start from functions, environments and duration before boxes.
- Mass, power, data and heat are shared.
- Cruise, manoeuvre, communication and safe mode activate different loads.
- Two units sharing power or software can fail together.

4 — Formulas, only when they answer a question

- How to read it : total mass equals the sum of item masses
- Σ means add every item in the budget.
- How to read it : total power equals sum of active loads
- The sum must be made by operating mode, not blindly across all hardware.

5 — What units and margins mean

- kg, W, Wh/J, bit/s and temperatures remain separate: different physical quantities are never added.

6 — Three concrete demonstrations, calculated step by step

- 420 kg structure + 180 kg avionics + 260 kg power + 640 kg payload.
- 10% reserve = 150 kg
- Total with reserve = 1,650 kg
- Cruise 900 W; manoeuvre adds 600 W.
- Check generation and distribution at peak.
- Useful computer power 80 W, converter 90%.
- Upstream = $80/0.90 = 88.9\text{ W}$

Active reading

Sections:	2 — Essential words, explained before using them ; 3 — See the architecture before calculating
Verification:	A value is usable only with its unit and calculation boundary.
Exercise:	change one assumption and predict the direction of change before recalculating

Spacecraft anatomy: bus, payload and subsystems

Those 8.9 W become local heat.

7 — Deepening: what the simplified diagram hides

A heater depends on sensor, software, relay and power.

Decide what stays alive when capability falls.

Analysis, inspection, test and demonstration close the loop.

8 — Application to an Earth-Mars spacecraft

9 — Reference dossier: what a real project must still consider

A spacecraft should not be designed by first choosing a battery, computer or engine. Start with the mission: payload, destination, duration, navigation accuracy, communications, environment, autonomy, safety and possibly crew. These needs become measurable requirements. Functional architecture describes what the vehicle must do; physical architecture later assigns those functions to real hardware. Keeping those two levels separate prevents the mission from being forced into a collection of preselected boxes.

The bus is not necessarily one physical object. It groups common support functions that make the mission possible. Payload is what directly accomplishes the primary objective. On an Earth-Mars vehicle the boundary can become ambiguous, so the important issue is not the label but clear interfaces, ownership, budgets and modes. A poor boundary can hide a critical dependency.

Mass, power, energy, volume, heat rejection, data rate, propellant and computing time are shared resources. A budget is therefore not an end-of-project spreadsheet. It prevents one team from optimising locally at the expense of the vehicle. A heavier instrument affects structure and propulsion; a stronger transmitter affects power and thermal control. Margins must be defined rather than assumed.

Launch, cruise, high-rate communications, manoeuvres, charging, science, emergency and safe mode do not activate the same equipment. Adding every load can over-size a system, while using only an average can hide an impossible peak. A mode-function matrix makes required loads, allowed shutdowns, redundancy and transitions explicit. For Mars, surviving a long degraded mode may matter more than a short peak capability.

Two computers sharing one converter are both lost if that converter fails. Identical software can share the same design defect, and co-located sensors can share the same environmental hazard. Robust design therefore considers common-cause failures and may separate power feeds, data paths, sensors, locations or technologies.

Mars missions change the value of accessibility, spares, tools, connectors, procedures and fault isolation. A compact design may save mass yet make repair impossible. A modular design may cost mass but improve maintainability. The correct trade depends on mission duration, crew, autonomy and consequence of failure.

Active reading

Sections: 7 — Deepening: what the simplified diagram hides ; 8 — Application to an Earth-Mars spacecraft

Verification: A value is usable only with its unit and calculation boundary.

Exercise: change one assumption and predict the direction of change before recalculating

Spacecraft anatomy: bus, payload and subsystems

A neat diagram is not proof. Important requirements need verification by analysis, inspection, demonstration or test, interfaces must be exercised, degraded modes must be tested and margins must be recalculated after changes. Validation asks a different question: even if the design meets its requirements, does it actually satisfy the mission need?

10 — Common traps and bad intuitions

Adding powers without a mode.

Bus, payload and interfaces: treating the spacecraft as one system

The bus contains functions that let the vehicle exist and operate: structure, power, thermal control, avionics, communications, attitude and often propulsion. Payload is what produces the mission effect. The boundary is primarily an engineering responsibility boundary that allows requirements and interfaces to be traced.

An interface turns a local subsystem into a system problem. A camera needs mass and mounting but also power, data, thermal rejection, field of view, processor time and operational procedures. Budgets must be checked by mission mode because a daily average can hide an impossible peak.

Redundancy must also be judged at interfaces. Two computers powered by the same converter or running the same faulty software may share a common cause. Architecture is therefore about organizing dependencies, not only selecting capable components.

11 — Guided exercises

- Question : What question comes before choosing hardware?
- Question : Why is a nominal result insufficient?

12 — What to remember

- Explain the topic in simple words before symbols.
- Connect at least four interfaces with other subsystems.
- Redo the three numerical examples without reasoning gaps.

13 — NASA sources for further study

- NASA - 2026 State-of-the-Art: Complete Spacecraft Platforms
- NASA - 2026 State-of-the-Art of Small Spacecraft Technology

Active reading

Sections:	10 — Common traps and bad intuitions ; Bus, payload and interfaces: treating the spacecraft as one system
Verification:	NASA - 2026 State-of-the-Art: Complete Spacecraft Platforms NASA - 2026 State-of-the-Art of Small Spacecraft Technology NASA Systems Engineering Handbook
Exercise:	change one assumption and predict the direction of change before recalculating