

AM-08.04 — IMU, gyroscopes and accelerometers: sensing motion without GPS

How can a vehicle estimate rotations and velocity changes when no GPS gives it a position?

AM-08.04 · SPACE ACADEMY

The IMU senses internal motion



Delta-Sierra · Space Academy

An IMU provides fast rates; other sensors are needed to limit drift.

1 - Understand the scene

A spacecraft crosses a region where no external beacon is available. It does not become blind: it carries internal sensors. A gyroscope measures rotation; an accelerometer measures specific acceleration associated with non-gravitational forces. Several such sensors are commonly packaged in an IMU.

The advantage of inertial sensing is autonomy and high update rate. Its weakness is equally important: small measurement errors are integrated over time. Without external updates, the estimate gradually drifts.

2 - Essential vocabulary

IMU - Inertial Measurement Unit, usually combining multi-axis gyroscopes and accelerometers.

Gyroscope - A sensor measuring angular rate about an axis.

Accelerometer - A sensor measuring specific force: non-gravitational force per unit mass.

Bias - A small approximately constant offset in a measurement.

Drift - Error that grows over time when measurement errors are integrated.

Integration - Accumulation of a rate over time: acceleration to velocity, velocity to position.

3 - Step-by-step mechanism

Three axes to describe space

An IMU commonly has three rotation axes and three acceleration axes. Their orientation relative to the vehicle must be known or the measurements cannot be interpreted correctly.

From angular rate to attitude

A gyro may report 2°/s about one axis. If the rate stays constant for 5 s, the angle changes by about 10°. Flight software repeats this reasoning at high frequency with changing measurements.

From acceleration to velocity

An accelerometer measures a rate of velocity change. Accumulating it over time estimates a velocity change. A second integration can estimate position change, but every integration magnifies bias effects.

4 - Formulas and reading

$$\Delta\theta \approx \omega \times \Delta t$$

"delta theta is approximately omega times delta t." Omega is angular rate here.

Over a short interval with nearly constant angular rate, rate times duration gives angle change.

$$\Delta v \approx a \times \Delta t$$

"delta v is approximately a times delta t."

For a short nearly constant acceleration interval, acceleration times duration gives the velocity change.

5 - Units and checks

Gyros commonly use degrees per second or radians per second. Accelerometers use m/s², sometimes multiples of g. An acceleration bias may look tiny, but its position effect grows quickly after two integrations.

Do not confuse 'IMU-measured acceleration' with gravitational acceleration: an accelerometer in free fall can read near zero even though gravity acts on the vehicle.

6 - Three demonstrations

Example 1 — Simple rotation

The gyro reports 3°/s for 4 s about one axis.

$$\begin{aligned}\Delta\theta &\approx 3 \times 4 \\ \Delta\theta &\approx 12^\circ\end{aligned}$$

Conclusion: The vehicle rotated about 12° if the rate stayed constant.

Example 2 — Measured burn

A thruster produces an average 0.5 m/s² acceleration for 20 s.

$$\begin{aligned}\Delta v &\approx 0.5 \times 20 \\ \Delta v &\approx 10 \text{ m/s}\end{aligned}$$

Conclusion: The IMU can help detect that the target velocity change has been reached.

Example 3 — Small bias, large drift

Assume a simplified constant acceleration bias of 0.01 m/s² for 100 s with no external update.

$$\begin{aligned}\text{velocity error} &\approx 0.01 \times 100 = 1 \text{ m/s} \\ \text{position error} &\approx 1/2 \times 0.01 \times 100^2 \\ \text{position error} &\approx 50 \text{ m}\end{aligned}$$

Conclusion: This deliberately simple model shows why inertial estimates need external correction.

7 - Deepening

“Inertial” does not mean infallible

An IMU needs no external signal for its instantaneous readings. That makes it valuable during occultations or rapid manoeuvres, but it does not automatically know absolute position.

Calibration before and during flight

Bias, scale factor, axis misalignment, and temperature dependence are characterized. Flight software can estimate some errors when other sensors provide a reference.

Saturation and dynamics

Every sensor has a measurement range. Excessive rotation or acceleration can saturate it; a very wide range may reduce useful resolution for small changes.

Why sensors are fused

The IMU provides rapid continuity; star trackers, radio tracking, or vision provide slower absolute references. Fusion combines strengths rather than asking one sensor to do everything.

8 - Why this matters for Mars

An interplanetary burn, atmospheric descent, or docking manoeuvre cannot wait minutes for Earth. The IMU provides rapid information needed by control loops.

On Mars, rovers and aerial vehicles can combine inertial sensing, vision, and maps to keep navigating when Earth communications are slow or unavailable.

9 - Common traps

- Thinking an accelerometer measures velocity directly.
- Thinking an IMU automatically provides absolute position.
- Ignoring bias: a small integrated error becomes large.
- Confusing specific force with gravity.
- Integrating measurements without expressing them in the correct frame.

10 - Guided exercises

Question: A gyro reads $1.5^\circ/\text{s}$ for 8 s. What simplified angle change?

Answer: $1.5 \times 8 = 12^\circ$, assuming constant rate.

Question: An acceleration of 0.2 m/s^2 lasts 15 s. What Δv ?

Answer: $0.2 \times 15 = 3 \text{ m/s}$.

Question: Why can a camera correct IMU drift?

Answer: Because observed environmental features provide an external reference that can reset the accumulated inertial estimate.

11 - What you should explain

- Define IMU, gyroscope, accelerometer, bias, and drift.
- Convert angular rate to a simple angle change.
- Convert acceleration to a simple Δv .
- Explain why integration accumulates errors.
- Explain why IMUs are combined with external references.

12 - NASA / JPL sources

NASA - State-of-the-Art of Small Spacecraft Technology, Guidance Navigation and Control
<https://www.nasa.gov/smallsat-institute/sst-soa/guidance-navigation-and-control/>

Free educational resource: this lesson explains the principles for beginners. It does not replace operational mission design, flight models, procedures, testing, or agency validation.